#### CLOSED CHAMBER WELL TEST

# INCLUDING

# FRICTIONAL EFFECTS

# **A** REPORT

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## ABSTRACT

Frictional effects were included in the closed chamber well test model in order to develop a more general solution for the closed chamber well test. Superposition of the cumulative influx, constant pressure solution of the radial diffusivity equation is used to overcome the limitations and difficulties resulting from solving the diffusivity equation in the presence of changing wellbore storage and frictional effects.

A sensitivity study was performed to analyze the influence of different tool and reservoir parameters on the closed chamber well test in the presence of frictional effects. Frictional effects significantly affect the early time pressure response of the closed chamber well test.

The superposition model was also improved by using variable time steps, hence, increasing the computation efficiency of the model.

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## **1. INTRODUCTION**

The closed chamber well test is a suitable **method** to identify and Sample formation fluids, **as** well as to determine oil reservoir parameters required for estimating productivity.

Closed chamber well testing is used in the petroleum industry in the **form** of backsurge perforation cleaning (*Simmons* (1985), *Simmons* (1986)). As shown in Figure 1, the equipment used for backsurge operations includes: a work string composed of two remote controlled valves, a temporary packer, and a pressure recorder. The assembly is run into the wellbore with an enclosed chamber formed between the upper and the lower surge valves. The increase in the hydrostatic pressure is recorded as the assembly is run into the well. When the packer is set the completion fluid overbalance is relieved, and the bottom hole pressure becomes equal to the static initial reservoir pressure. When the lower valve is opened the drawdown is obtained, as the formation sandface is exposed to a minimum pressure, and fluids are produced. Then, the fluid level rises and the bottom hole pressure increases until it reaches the static reservoir pressure. Finally, the upper valve *is* opened, the packer released and the bottom hole pressure returns to an overbalance.

The closed chamber well test is similar to **a** conventional drillstem test; moreover, it is a generalized form **of** the drillstem test known **as** slug test. The closed chamber well test involves liquid level changes in the wellbore as a result **of** the instantaneous removal of a specific amount of liquid **frcm** the wellbore. The main difference between the closed chamber test and the slug test is wellbore storage. The slug test wellbore storage is constant, related either **to** fluid level rise or to fluid compression in a fixed volume. The closed chamber test wellbore storage varies **from** being controlled by fluid level rise, to fluid compression in a changing volume. **In** a closed chamber test, the initial wellbore storage is high and reduces **dur**ing the test, **as** the chamber gas compresses above the liquid column.

The variable wellbore storage and the presence of frictional and momentum effects throughout the test, make the closed chamber well test problem non-linear and difficult solve

analytically. Hence, numerical techniques **are** required to overcome the limitations and difficulties resulting from these non-linearities.

This study was performed in order to develop a more general solution for the closed chamber well test by including frictional effects in the closed chamber well test model developed by *Simmons* (1985) as well as improving the efficiency of the model by using variable time steps.

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Figure 1: Schematic Representation of the Closed Chamber Equipment (after Simmons (1985), Simmons (1986))

#### 2. LITERATURE SURVEY

The first application of slug test analysis was presented by *Ferris and Knowles* (1954), who proposed the instantaneous slug test method for determining the transmissivity of an aquifer. They analyzed late time data considering the instantaneous line source response, but neglected the effects of wellbore storage and **skin**.

*Cooper, Bredehoeft, and Papadopulos* (1967), presented a solution for the variation in the liquid level when a slug test **is** performed. They considered wellbore storage in the analysis, and proposed using type curves for determining reservoir transmissivity.

After that, *Ramey and Aganval* (1972) presented the slug test solution including the effects of wellbore storage and skin, but momentum, friction, phase change and wellbore fluid compressibility effects were neglected. *Ramey, Agarwal, and Martin* (1975), presented type curves for the slug test. Three kinds of curves were developed: a) log-log early time, b) semilog intermediate time, and c) log-log late time. These curves combine skin effect as well as wellbore storage into a correlating parameter that may be determined from a type curve match.

A theory for analyzing closed chamber well **tests** was presented by *Alexander* (1977) and *Marshall* (1978). They suggested using the results obtained by performing a closed chamber test to monitor the initial flow period of the drillstem test, **as** well **as** to identify and measure formation properties.

As an extension of the general slug test solution, *Shinohara* (1980), presented a mathematical solution to **analyze** data from closed chamber tests. The solution he proposed can be applied for **the** closed chamber test problem except when the volume of the closed chamber **is** very **small**, or there is a considerable difference between the **initial** pressure of the closed chamber and the atmospheric pressure. The solution was obtained based on a wellbore momentum balance. Some of the assumptions adopted in the analysis were that the wellbore friction, the compressibility of the liquid in the wellbore, and **mass** transfer between liquid and **ges** were negligible. Also, ideal chamber gas behavior was assumed.

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After that, *Saldana* (1983) proposed a mathematical solution to describe the flow phenomena occurring during a slug test, **a** drillstem test, or a closed chamber test. The solution was obtained by applying a momentum balance equation to **the** liquid in the wellbore including gravitational, inertial, and frictional effects on the fluid column. *Saldana* **also** assumed ideal gas behavior in the analysis.

**Simmons (1985)** developed a closed chamber test model, obtained by **the** superposition of the constant pressure cumulative influx solution **to the** radial diffusivity equation. In his approach he considered real **gas** compressibility effects, but the effects of friction and momentum were not included. In addition, no mass exchange and incompressible wellbore liquids were assumed.

Based on the results of Simmons's approach, *Simmons and Sageev* (1985) presented a method for analyzing backsurge pressure data to determine the reservoir transmissivity and the Hurst skin effect. In a paper to be presented in 1986, *Simmons* presents a new method for analyzing closed chamber tests. In this method, the recorded chamber pressures are differentiated, yielding the instantaneous sandface rate. Then, the superposition model of constant rates is invoked allowing the determination of formation transmissivity, and the variation of wellbore skin with time. However, we still lack a method for reproducing the bottom hole pressure including the frictional and momentum effects.

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#### 3. CLOSED CHAMBER PRESSURE RESPONSE ANALYSIS

# 3.1. Fluid Flow in Pipes

The pressure drop for a fluid flowing between two points, (1 and 2), in a pipe is expressed by:

$$144\left[p_{1}v_{1}+\int_{v_{1}}^{v_{2}}pdv-p_{2}v_{2}\right]=\frac{g}{g_{c}}\left(z_{2}-z_{1}\right)+\alpha\frac{\left(V_{2}^{2}-V_{1}^{2}\right)}{2g_{c}}+h_{f}\frac{g}{g_{c}}$$
(1)

where:

p =pressurev =specific volume  $(v = 1/\rho_f)$ z =elevationV =average fluid velocity $h_f =$ frictional head lossa =kinetic energy correction termg =acceleration of gravity

 $8_c =$  conversion factor

For liquids, assuming that the density remains **constant**, equation (1) becomes the generalized Bernoulli equation for flowing liquids in a pipe:

$$144 \ \frac{p_1 - p_2}{\rho_f} = \frac{g}{g_c} \ (z_2 - z_1) + \alpha \ \frac{(V_2^2 - V_1^2)}{2g_c} + h_f \ \frac{g}{g_c}$$
(2)

where:

 $\rho_f =$ fluid density

Equation (2) can be expressed as

$$-144 \frac{\Delta p}{\rho_f} = \frac{g}{g_c} \Delta z + \frac{\alpha \Delta (V^2)}{2g_c} + \frac{g}{g_c} h_f$$
(3)

where:

 $\Delta p =$  pressure change

 $\Delta(V^2)$  = change in velocity terms

$$\Delta z =$$
 change in elevation

The kinetic energy correction factor,  $a_r$  is usually set equal to unity (*Benedict*, 1980). Then, when the pipe diameter is constant, the total pressure drop *can* be calculated from equation (3) as

$$-(\Delta p)_{total} = \frac{1}{144} \text{ Pf } \frac{g}{g_c} \Delta z + \frac{1}{144} \text{ Pf } \frac{g}{g_c} h_f$$
(4)

Equation (4) is of the general form:

$$-(\Delta p)_{lotal} = (\Delta p)_{elevation} + (\Delta p)_{friction}$$
(5)

## 3.2. Frictional Head Loss

The head loss due to friction in a pipe is defined by the Darcy-Weisbach equation as

$$h_f = f \frac{L}{D} \frac{V^2}{2g} \tag{6}$$

where:

f = friction factor

t

- L = length
- D = inside pipe diameter

The frictional pressure drop in Equation (5) is then given by

$$(\Delta p)_{friction} = \frac{1}{144} \rho_f \frac{g}{g_c} h_f = \frac{1}{144} \rho_f f \frac{L}{D} \frac{V^2}{2g_c}$$
(7)

Expressing (6) as a function of the flow rate, q, we have

$$h_f = 8 f \frac{L q^2}{\pi^2 D^5 g}$$
(8)

Substituting Equation (8) in Equation (7), yields the frictional pressure **drop** expressed as **a** function of **the** flowrate

$$(\Delta p)_{friction} = \frac{8}{144} \rho_f f \frac{L q^2}{\pi^2 D^5 g_c}$$
(9)

The friction factor, f, as presented by *Moody* (1944) is a function of two dimensionless parameters:

• the relative roughness, e/D, where e is a dimensional quantity that represents the absolute roughness, and

• the Reynolds number,

$$\operatorname{Re} = \frac{\rho_f V D}{\mu}$$
(10)

where  $\mu$  is the viscosity of the fluid in *lbm/ft–sec*.

The Reynolds number is dimensionless but may be expressed in oilfield units as

$$Re = \frac{13033 \, q_o(bbl/D)}{\mu_o(cp) \, D(inches) \, (131.5 + API)}$$
(11)

The Moody friction factors are also defined according **to** the flow regime:

• Laminar zone (0  $c \text{ Re} \leq 2000$ )

$$\mathbf{f} = \mathbf{64}/Re \tag{12}$$

• Critical zone (2000 < Re ≤ 4000)

$$f = 0.5/\text{Re}^{0.3}$$
 (13)

• Transition zone (4000 < Re  $\leq$  (200*D*/*e*)<sup>1.16</sup>)

$$1/\sqrt{f} = 1.14 - 2\log(e/D + 9.34/Re\sqrt{f})$$
(14)

• Turbulent zone (Re > (200*D*/*e*)<sup>1.16</sup>)

$$1/\sqrt{f} = 1.14 - 2\log(e/D) \tag{15}$$

An iterative calculation is required to evaluate the Moody friction factor in the transition zone (Equation (14)).

#### 33. Closed Chamber Well Test Model

#### **3.3.1.** Without Frictional Effects

Simmons (1985) developed a model for analyzing the pressure response for the closed chamber well test. The model presented by *Simmons* uses superposition, hence, simplifying the solution of the diffusivity equation in the presence of changing wellbore storage. The net influx at a given time **step is** calculated by using superposition of the cumulative influx, constant pressure solution of the radial diffusivity equation.

The dimensionless radial diffusivity equation is expressed by

$$\frac{\partial^2 p_D}{\partial r_D^2} + \frac{1}{r_D} \frac{\partial p_D}{\partial r_D} = \frac{\partial p_D}{\partial t_D}$$
(16)

where the dimensionless time,  $t_0$ , and dimensionless radius,  $r_D$ , are defined as

$$t_D = \frac{73.25 \times 10^{-9} \, kt}{\phi \mu c_t \, r_w^2} \tag{17}$$

$$r_D = \frac{r}{r_w} \tag{18}$$

and the dimensionless pressure,  $p_D$ , for a constant pressure inner boundary is given by

$$p_D = \frac{p_i - p_{wf}}{p_i - p_o} \tag{19}$$

The assumptions considered in **the** development **of the** model were:

- **a.** There **is** no mass exchange between the produced **liquids** and the chamber gas.
- **b.** Incompressible wellbore liquids.

t

c. Negligible momentum and frictional effects.

- d. The flowrate during the test period is not impeded by critical flow.
- e. Only liquid is produced from the reservoir during the test.
- f. The reservoir behaves as an infinite homogeneous radial system of isotropic properties during the test period.

g. Negligible gradients of pressure and temperature with respect to depth in the gas column.

The closed chamber model discretizes the pressure response into constant pressure intervals and assumes that the pressure at the beginning of the time step remains constant during the time step. By using superposition of **the** cumulative influx, constant pressure solution of the radial diffusivity equation, the net influx after N time steps is calculated **as** 

$$N_{p} = \beta \left[ p_{i} - p_{o} \right] Q_{D}(N \Delta t_{D}) - \beta \sum_{j=1}^{N-1} \left[ \left[ p_{j} - p_{j-1} \right] Q_{D}(\left[ N-j \right] \Delta t_{D}) \right]$$
(20)

where  $N_p$  is the fluid produced (in *bbls*) after **N** time steps,  $\Delta t_D$  is the dimensionless time step, evaluated based on the time step At, and  $\beta$  is a constant of proportionality equal to

$$\beta = 1.119 \phi h c_l r_w^2 \tag{21}$$

 $\mathcal{Q}_D$  is the dimensionless cumulative influx defined as

$$Q_D = \frac{Q}{1.119\phi hc_i r_w^2 (p_i - p_{wf})}$$
(22)

and is evaluated by inverting the dimensionless Laplace solution presented by Da Prat (1981)

$$\overline{Q}_{D} = \frac{\sqrt{s} K_{1} (6)}{s^{2} [K_{0}(\sqrt{s}) + S \sqrt{s} K_{1}(\sqrt{s})]}$$
(23)

where  $K_o$  and  $K_1$  are the zero and first order modified Bessel functions of the second kind.

From the cumulative influx, the fluid level **X**, shown in Figure 2, is calculated as

$$\mathbf{X} = L_{ci} + \frac{N_{p}}{A_{ch}} \tag{24}$$

where:

 $L_{ci}$  = initial fluid level

*A*<sub>*ch*</sub> = area of the chamber

and by assuming isothermal gas compression the chamber pressure is calculated as

$$p_{ch} = \frac{p_{ch_i} \left[ L_c - L_{ci} \right] Z}{\left[ L_c - X \right] Z_i}$$
(25)

where:

 $p_{ch_i}$  = initial chamber pressure

 $p_{ch}$  = chamber pressure

 $Z_i$  = initial compressibility factor

**Z** = compressibility factor

Assuming negligible momentum and frictional effects, the bottom hole pressure is calculated as

$$p_{wf} = p_{ch} + \frac{1}{144} \rho_f \frac{g}{g_c} X$$
(26)

where:

 $p_{wf}$  = bottom hole flowing pressure

 $p_{ch} =$  chamber pressure

and the second **term** in **the right** hand side is the hydrostatic pressure **of** the liquid column.



Figure 2: Definition of Variables for the Closed Chamber Model (after Simmons (1985), Simmons (1986))

#### **33.2.** With Frictional Effects

Starting with the closed chamber well test model presented in the preceding section, and neglecting momentum effects, **the** frictional effects can be included in the model. The friction head changes with time until the fluid stabilizes at the initial static liquid level. By using Simmons's model, assuming that the pressure at the beginning of **the** time step remains constant during the time step, and taking the average rate and the maximum value of **AX** per time step, the pressure response can **be** generated **for** each time step.

The frictional pressure drop is:

$$h_f = 8f \, \frac{(X + \Delta X) \, q^2}{\pi^2 D^5 g} \tag{27}$$

The oil flowrate changes during the test, and is a function of the bottom hole pressure. For the constant pressure solution of the radial diffusivity equation, the fluid influx rate at the end of a period of interest can be calculated by first determining the corresponding fluid influx terms and approximating their time derivative by dividing the difference between each two successive values by the corresponding difference in absolute time (*Chatas*, **1953**).

That means, the average oil flowrate during the time step j is:

$$q_j = \frac{N_{p_j} - N_{p_{j-1}}}{t_j - t_{j-1}}$$
(28)

Then, the bottom hole flowing pressure considering friction is:

$$p_{wf} = p_{ch} + \frac{1}{144} \rho_f \frac{g}{g_c} (X + \Delta X) + \frac{8}{144} \rho_f f \frac{(X + \Delta X) q^2}{\pi^2 D^5 g_c}$$
(29)

Equation (29) expresses the pressure response during the closed chamber well test, taking into account the frictional effects. The calculation of the pressure response by using the model presented in section 3.2., but including the frictional effects, implies that the assumptions esta-

blished in the development of the model also apply in this case.

## 3.3.2.1. Fixed Time Step

The computer program for the closed chamber well test model, including frictional effects, is presented in Appendix **1**. Generally, when using constant time steps, an extremely small time step is required to represent accurately the chamber pressure **rise** and to avoid numerical over shooting above the upper surge valve due to excessive variation of the fluid level. Therefore, to calculate the pressure response over a reasonable interval of time requires a large number of time steps; that implies, unreasonable amount of computer time.

**Simmons** recommended to improve the superposition model by using variable time **steps**, such that the amount of numerical computations be decreased. For this study, a new model with variable time steps was developed. This model is described in the next section.

#### 3.3.2.2. Variable Time Steps - Interpolator

To increase the efficiency of the superposition model a new program that uses variable time steps was developed. The computer program of the closed chamber model with variable time steps is presented in Appendix 1.

The new model allows the use of different time increments along the closed chamber well test. For instance, a larger time increment is used when the chamber pressure is insignificant compared to the reservoir pressure; then, when the fluid level approaches the upper surge valve the time increment is reduced to avoid over shooting above the upper surge valve; finally, larger time increments are used when the chamber pressure has increased close to **its** final pres**sure**.

The evaluation of the net influx at a given time by using superposition requires the calculation of the dimensionless cumulative influx (Equation 22) for all combinations of *the* previous time steps. When variable time steps are used, new dimensionless cumulative influx values have to be obtained for each pressure change in all the pressure history for every time **step. To** calculate these dimensionless values by inverting the dimensionless Laplace solution (Equation **23**) would be a disadvantage and the computational time will increase instead of decrease. To avoid **this** problem an interpolator was developed. **This** interpolator interpolates between two successive values in a given table that contains the dimensionless cumulative influx values for different dimensionless time values. Thus, for a given dimensionless time the interpolator computes the closest lower entry in the table, and interpolates between **this** and the following entry in the table. The interpolator subroutine is presented in Appendix **1**.

To check the new program, the closed chamber well test was analyzed for a specific case with the two models: with a fixed time **step** and with variable time steps. One **of** the cases **stu**died in the sensitivity study, that will **be** presented later in section **5**, was chosen. The selected case corresponds to a skin value **of 2** and zero initial fluid level.

Figures **3** and **4** show the comparison between the results obtained with the two models. Figure **3** presents the bottom hole flowing pressure response and Figure **4** presents the frictional pressure drop. The dotted Curves in both figures represent the response obtained with a fixed time step, and the continuous curves correspond to the response with variable time **steps**. According to these figures, the pressure response obtained with both models is identical. Moreover, **only** 1000 time steps were required with variable time steps, while 10000 time steps were used with a fixed time step (0.01 seconds in **this** case). That means, that using the new program reduced the number of computations and CPU time **by** a factor of 10.

The new model including variable time **steps is** an interactive program, in which the time steps **are** chosen **according** to the development of the pressure response. **This** model could be improved **by** allowing the computer to select the time step during the well test. Some important facts should **be** considered in such a model: the oil flowrate continuously decreases during **the** test, and the rate of change in the bottom hole flowing pressure **has** to **be** maintained within a tolerance range avoiding the over shooting above the upper surge valve.

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Figure 3: Bottom Hole Pressure Obtained with 2 Different Models



Figure 4: Friction Pressure Drop Obtained with 2 Different Models

#### 4. VERIFICATION OF THE MODEL INCLUDING FRICTION EFFECTS

#### 4.1. Verification of Basecase Results

Since the difference between the model presented in **this** study and Simmons's lies in the consideration of frictional effects, the new model can be tested by generating Simmons's basecase for no friction conditions in the tubing.

The basecase response that will be explained in next section was obtained by using an absolute roughness factor, **e**, of zero. In other words, it was assumed that the frictional pressure drop during the test **is** always zero. Figure **5** presents the bottom hole flowing pressure response obtained with the **two** models. The continuous line represents the pressure response generated with Simmons's model. The dotted line corresponds to the pressure response calculated with the new model, assuming no frictional effects. According to **this** figure, **both** responses **are** identical. Hence, the new model accurately duplicates Simmons's results when the frictional pressure **drop is** neglected in the calculation of the closed chamber pressure response.



Figure 5: Bottom Hole Pressure for Basecase Without Friction

#### 42. Basecase Analysis

In this section we describe the basecase for the parameter investigation that will follow. The basecase described by **Simmons** consisted of zero skin and 100 feet of initial fluid cushion in the well above the lower valve. Since in this study the momentum effects in the wellbore were not considered, the basecase was changed so that there is no initial fluid level in the wellbore above the lower valve. More than that, for this study the momentum effects of the fluid between the lower valve and the formation were not considered. In other words, the model assumes that the lower valve is positioned opposite the tested zone. All the other parameters such as chamber length, initial chamber pressure, and fluid gravity are the same as were presented by **Simmons**. Table 1 presents the parameters of the basecase.

Figure **6** presents the closed chamber pressure response for the new basecase, without initial fluid level, and without wellbore frictional effects. There are **three** curves in Figure **6**. The uppermost curve represents the bottom hole pressure response, and is the sum of the two other curves: the hydrostatic fluid pressure drop and the chamber pressure. The hydrostatic pressure drop curve represents the fluid level rise after the lower surge valve is opened. As the chamber **gas** compresses, **an** abrupt rise in the bottom hole flowing pressure is observed. The difference between the chamber pressure and the bottom hole flowing pressure is the hydrostatic pressure drop of **the** fluid column, neglecting momentum and frictional effects.

Figure 7 illustrates the closed chamber pressure response for the basecase, without initial fluid level, but including wellbore filectional effects. In this figure, in addition to the three curves **shown** in Figure 6, **a fourth** curve is present that represents the frictional pressure drop in the wellbore. The bottom hole flowing pressure in this case corresponds to the sum of the fictional pressure drop, the hydrostatic fluid pressure drop and the chamber pressure. **As** in the basecase without friction, the abrupt rise in the bottom hole pressure **is** due to the rapid compression of the chamber gas. The separation between the **bottom** hole pressure and the frictional pressure is given in **this case** by the sum of the hydrostatic pressure drop and the frictional pressure drop and the frictional pressure drop and the frictional pressure is given in **this case** by the sum of the hydrostatic pressure drop and the frictional pressure is given in **this case** by the sum of the hydrostatic pressure drop and the frictional pressure drop. After about **25** seconds, when the friction pressure drop reduces to zero,

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the difference is given only by the hydrostatic pressure drop like in the basecase without friction.

Figure 8 shows the bottom hole pressure response for the basecase with and without friction. As can be observed in this figure, during the first 20 seconds of the test the bottom hole pressure with friction is larger than the bottom hole pressure without friction. At this time the two curves overlap. After that, the bottom hole pressure without friction becomes larger than the bottom hole pressure with friction. The reason for this can be explained by observing Figures 9, 10, 11 and 12, which represent respectively: closed chamber pressures for both cases, hydrostatic fluid pressure drops for both cases, frictional pressure drop for the basecase with friction, and oil flowrates for both cases.

According to these figures, during the first 20 seconds of the test, the most important factor in the bottom hole pressure response is given by the elevation pressure as well as the friction pressure, which is proportional to the fluid level. During these 20 seconds both pressure drops are increasing. But, because of the friction effects, the fluid level rise is delayed in the case when friction is considered (Fig. 10). The chamber pressure, when no frictional effects are considered, starts its abrupt rising earlier than in the friction case (Fig. 9). After 20 seconds, the frictional pressure drop starts diminishing rapidly (Fig. 11) due to the decreasing oil flowrate (Fig. 12); the hydrostatic fluid pressure drop is practically constant and equal for both cases; and the chamber pressure is higher in the case without friction than with friction. All effects combined result in frictionless late time bottom hole pressures higher than the late time bottom hole pressures with friction.

Figures **13** and 14 represent the log-log early and late time responses for the basecase with **and** without friction. The log-log early and late time responses are represented as functions of the dimensionless variables defined in section **3 of this** study. The early time response **is** significantly affected by the presence **of** wellbore friction, while the late time responses **are** very similar **for** both cases as **can** be observed in Figure 14.

# TABLE 1. BASECASE PARAMETERS

Parameter	Value	
Chamber Diameter	2.441	inches
Roughness	0.00060	inches
Relative Rouehness	0.00025	
Total Chamber Length	1000	feet
Initial Fluid Height	0	feet
Initial Chamber Pressure	30	psig
Chamber Gas Gravity	0.65	(to air)
Initial Reservoir Pressure	5000	psig
Reservoir Temperature	175	F
Produced Fluid Specific Gravity	25	API
Produced Fluid viscositv	1.25	CD
Porosity	27	%
Reservoir Permeability	100	md
Skin	0	
Well Diameter	10	inches
Formation Total Compressibility	10 x 10 <sup>-6</sup>	psi-'
Formation Thickness	25	feet
Total Test Time	100	seconds



Figure 6: Basecase Without Friction



Figure 7: Basecase With Friction



Figure 8: Bottom Hole Pressure for Basecase With and Without Friction



Figure 9: Chamber Pressure for Basecase With and Without Friction



Figure 10: Elevation Pressure for Basecase With and Without Friction



Figure 11: Frictional Pressure Drop for Basecase



Figure 12: Oil Flowrate for Basecase With and Without Friction



Figure 13: Early Time Plot for Basecase With and Without Friction



Figure 14: Late Time Plot for Basecase With and Without Friction

#### 5. SENSITIVITY STUDY

In this section we present a sensitivity study of various tool and reservoir parameters of the closed chamber test, including frictional effects.

#### 5.1. Effect of Wellbore Skin

Skin effect values of 0, 2 and 5 were considered in the sensitivity study. Figure 15 presents the bottom hole pressure response, including Friction effects, for the skin values considered in the analysis. Since the presence of wellbore skin reduces the sandface flowrate into the wellbore, the rapid pressure rise during chamber compression is delayed. Yet, the final fluid level and the late time responses are **similar**. The frictional pressure drops for these skin values are illustrated in Figure 16. According to this figure, the frictional pressure drop is smaller but lasts longer as the skin value increases. The reason for this is explained in Figure 17, that represents the oil flowrate for the skins studied. For higher skins the oil flowrate is smaller and decreases slower than for lower skins. In other words, because the cross sectional area of the pipe remains constant, and the initial fluid conditions are identical the fluid rises to the same level in all cases.

Log-log early and late time plots are presented in Figures **18** and **19**. Like in the cases studied by **Simmons**, the late time format gives the largest resolution to skin effect. The early time response in the presence of wellbore skin differs significantly **from** the early time response without wellbore skin as shown by *Sageev* (**1986**). In the early time log-log format the **slope** of the pressure response without wellbore skin is **1/2**, denoting transient linear flow. The early time response with wellbore skin has a unit slope, representing constant flow rate.

The bottom hole pressure responses, with and without friction, for skin values of 0, 2 and 5, are presented in Figures 8, 20 and 21 respectively. Because the frictional pressure drop decreases as the skin value increases, the difference between both curves becomes less significant for larger skins.

By observing the dimensionless plots of the pressure response, with and without friction, for the different skins analyzed (Figs. 13, 14, 22, 23, 24 and 25), we conclude that the late time plot matches better both responses. Moreover, for larger skins the difference between the curves is insignificant.


Figure 15: Bottom Hole Pressure for Different Skins



Figure 16: Frictional Pressure Drop for Different Skins



Figure 17: Oil Flowrate for Different Skins



Figure 18: Early Time Plot for Different Skins



Figure 19: Late Time Plot for Different Skins



Figure 20: Bottom Hole Pressure for Skin=2 With and Without Friction



Figure 21: Bottom Hole Pressure for Skin=5 With and Without Friction



Figure 22: Early Time Plot for Skin-2 With and Without Friction



Figure 23: Early Time Plot for Skin=5 With and Without Friction



Figure 24: Late Time Plot for Skin=2 With and Without Friction



Figure 25: Late Time Plot for Skin=5 With and Without Friction

## 5.2. Effect of Initial Reservoir Pressure

The values for initial reservoir pressure considered in the analysis were **3000** and 5000 psig, where the last one corresponds to the basecase.

Figure 26 shows the influence of the initial reservoir pressure on the bottom hole pressure when considering wellbore frictional effects. **As** expected, the bottom hole pressure response tends to the initial static reservoir pressure. Like in the cases presented by *Simmons*, the period of rapid pressure rise due to gas compression occurs earlier for higher pressure formations.

As can be observed in Figure 27, the frictional pressure drop is smaller and lasts longer **as** the initial reservoir pressure decreases. The reason for this is that the oil flowrate for higher pressure formations is larger because the greater initial pressure differential at the sandface (Figure 28). However, since the final volume of oil produced is about the same, (chamber **gas** is compressed **to** different pressures), the areas under the flowrate curves (Figure 28) are about the same.

Log-log early and late time plots for different initial reservoir pressures are presented in Figures 29 and **30.** Again, since skin is not present, the early time response is similar to the slug test response with a slope of **1/2**. The late time log-log pressure response approaches the unit slope of the line source slug test response presented by *Ferris and Knowles* (1954).

Figure **31** illustrates the bottom hole pressure response, with and without friction, for the case when **the** initial reservoir pressure is 3000 psig. Comparing this figure to the figure corresponding to the basecase (Figure **8**), it *can* be inferred that the relative effect of friction in lower initial reservoir pressure **formations** is smaller than in higher pressure formations.

Figures **32** and **33** present the early and late time plots **for an** initial reservoir pressure of **3000** psig, with and without friction and without wellbore skin. For smaller initial reservoir pressures the **late** time portion collapses better because the frictional effects in these cases are smaller.



Figure 26: Bottom Hole Pressure for Different Initial Reservoir Pressures



Figure 27: Frictional Pressure Drop for Different Initial Reservoir Pressures



Figure 28: Oil Flowrate for Different Initial Reservoir Pressures

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Figure 29: Early Time Plot for Different Initial Reservoir Pressures



Figure 30: Late Time Plot for Different Initial Reservoir Pressures



Figure 31: Bottom Hole Pressure for  $p_i$ =3000 psig with and without friction



Figure 32: Early Time Plot for  $p_i=3000$  psig with and without friction



Figure 33: Late Time Plot for  $p_i$ =3000 psig with and without friction

### 53. Effect of Chamber Diameter

The closed chamber pressure response was studied for two chamber diameters, **2.441** and **4.0** inches. The closed chamber pressure response when considering frictional effects in the wellbore is closely related to the chamber diameter because the frictional head loss **is** inversely proportional to  $D^5$ .

Figure **34** presents the bottom hole pressure for the two cases studied. **As** the tubing **di**ameter increases, the rapid chamber compression is delayed. The frictional pressure drops for these cases are represented in Figure **35. As** the chamber diameter is increased the friction pressure losses become less important in the behavior of the total pressure response. For example, according to Figure **36**, for a chamber diameter of **4.0** inches, the friction pressure loss is practically negligible though it lasts longer, and both pressure responses, with friction and without it, are practically identical.

The oil flowrate for both cases of Figure **34** are shown in Figure **37**. As expected, the area beneath the curve for a chamber diameter of **4.0** inches is larger than the area beneath the curve corresponding to **a** chamber diameter of **2.441** inches, since wellbore volume is larger. Like in the no-friction cases presented by *Simmons*, there is a dimensionless time shift to the right on the early and late time log-log pressure responses due to larger chamber diameters (Figures **38** and 39).

Early and late time plots for the **case** corresponding to a chamber diameter of 4.0 inches can be observed in Figures 40 and 41, Again, for a larger chamber diameter, because the frictional pressure drop is smaller, the late time log-log responses are practically identical. The early time **log-log** responses **are** slightly different during the "slug test" response. However, during the rapid pressure rise and thereafter the responses are practically identical



Figure 34: Bottom Hole Pressure for Different Chamber Diameters



Figure 35: Frictional Pressure Drop for Different Chamber Diameters



Figure 36: Bottom Hole Pressure for  $D_{ch}$ =4.00 inches With and Without Friction



Figure 37: Oil Flowrate for Different Chamber Diameters



Figure 38: Early Time Plot for Different Chamber Diameters



Figure 39: Late Time Plot for Different Chamber Diameters



Figure 40: Early Time Plot for  $D_{ch}$ =4.00 inches With and Without Friction

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Figure 41: Late Time Plot for  $D_{ch}$ =4.00 inches With and Without Friction

### 5.4. Effect of Roughness

Because the Moody friction factor is a function of **the** relative roughness, e/D, we analyzed the influence of **this** parameter on the closed chamber pressure response. Two different values of the absolute roughness, e, were used in the sensitivity study.

A typical absolute roughness value selected for the basecase was 0.0006 inches. The corresponding relative roughness, for a chamber diameter of **2.441** inches, **is** 0.00025. To perform the analysis the another value considered for the absolute roughness was 0.00015 inches, that gives a relative roughness of 0.00006 for the same chamber diameter. It is important to notice that a skin value of **2** was used to generate the pressure response for the different relative roughness values.

Figures 42, 43 and 44 present the results obtained. According to these figures, although the frictional pressure drop is lower for the smallest relative roughness, the bottom hole pressure response is not significantly altered. Also, the decreasing of the flowrate behaves in the same way for the two cases studied. The reason for this is that the flow regime in both cases lies in the transition zone where Moody friction factors do not vary much for different values of e/D.



Figure 42: Bottom Hole Pressure for Different Roughness



Figure 43: Frictional Pressure Drop for Different Roughness



Figure 44: Oil Flowrate for Different Roughness

### 55. Effect of Initial Chamber Pressure

Two values of initial chamber pressure were considered: 0 and **30** psig (basecase). Again, the closed chamber pressure responses for these two values were generated for a skin value of 2. Figure **45** illustrates the bottom hole flowing pressure response for the two different values of initial chamber pressure. The frictional pressure drops for these are presented in Figure **46**. Like in the no-friction cases studied by **Simmons**, larger initial chamber pressure values tend to smooth the pressure response. **Also**, as expected, the frictional pressure drops for these are smaller because less fluid rise is required, and the flowrates are smaller er.

Early and late time dimensionless plots are presented in Figures **47** and **48**. Similar to the no-friction cases there is a **shift** to the right in the early time plot for lower initial chamber pressures, and the late time response is **not** significantly affected by the initial chamber pressure.



Figure 45: Bottom Hole Pressure for Different Initial Chamber Pressures



Figure 46: Frictional Pressure Drop for Different Initial Chamber Pressures



Figure 47: Early Time Plot for Different Initial Chamber Pressures

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Figure 48: Late Time Plot for Different Initial Chamber Pressures

## 6. CONCLUSIONS AND RECOMMENDATIONS

In this study, frictional effects were included in the closed chamber well test model developed by *Simmons* (1985) in order to develop a more general solution for the closed chamber well test.

According to the results obtained in this study, frictional effects significantly affect the early time pressure response, while late time pressure responses are not highly affected. The inclusion of tubing frictional losses in the closed chamber test acts like a choke, and reduces the instantaneous flowrate into the wellbore. However, at early time, when the bottom hole pressure is not affected by the pressure in the chamber, the presence of frictional effects increase the bottom hole pressure in comparison to the frictionless case. The gas compression in the chamber **is** only a function of the liquid level, (no mass transfer between the liquid column and the chamber **gas**). Since the flowrate is restricted by the tubing friction, the liquid levels are lower than in the frictionless case, and the rapid compression of the chamber gas is delayed. Hence, the late time bottom hole pressure for the friction case is lower than the bottom hole pressure for the frictionless case.

A sensitivity study was performed to analyze the influence of different tool and reservoir parameters on the closed chamber well test including frictional effects. The frictional pressure drop is closely related to the tool parameters like chamber diameter, absolute roughness and chamber length. Frictional losses could be reduced by controlling these parameters. For in-**stance**, for large chamber diameters the friction pressure losses become small enough such that the total pressure response **is** practically unaffected.

In general, the frictional pressure loss during a closed chamber test increases rapidly, levels off **as** it goes through a maximum, and finally decreases rapidly **as** the chamber gas compresses and chokes the well. We observe that as the frictional effects increase, the dura**ticns of** significant frictional head losses decrease. **This** reduction in the durations of the **frictional** effects is observed when we compare closed chamber **tests** with different diameters, different initial **reservoir** pressures **and** different wellbore **skins**.

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As expected, the sensitivity study showed that the early time responses in the presence of wellbore skin differ significantly from early time responses without wellbore skin. In log-log format the early time response without wellbore skin is similar to the slug test response with a **slope** of 1/2 denoting transient linear flow. On the other hand, the early time log-log response with wellbore skin has a unit slope, representing constant flowrate. The flowrate profile in the presence of high wellbore skin is practically constant, and decreases rapidly like a shut-in. Hence, the late time build-up could be approximated as a simple buid-up flowing a constant rate flow period, and analyzed using a Homer technique.

The closed chamber well test model developed by **Simmons** was also improved by using variable time steps instead of a fixed time step. With variable time steps the computer time is reduced significantly because the number of computations is reduced by an order of magnitude in comparison to the calculations required when using fixed time steps. The program with variable time steps should be improved by including a restarting routine such that the program could be restarted from a given time greater than **zero**.

Momentum effects of the fluid between the lower valve and the formation, and in the cushion above the lower surge valve, were not considered in the model. Momentum effects are important in the cases where high flowrates occur and the initial fluid column in the wellbore must accelerate rapidly. **For** this reason it is recommended to include momentum effects on the closed chamber well test.

In developing the model, it is also assumed that wellbore liquids are incompressible. The late time response depends significantly on the high pressure compressibility of the gas. Yet, at these high pressures, the volume of the gas is small. On the other hand, the liquid column compressibility is low, but its volume is large. The wellbore storage is a function of the two fluid columns in the wellbore. Hence, future studies should analyze the influence of the produced liquid compressibility on the closed chamber pressure response.

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# Summary

- Frictional effects were included in the superposition model for the closed chamber well test.
- **2.** Frictional effects significantly affect the early time pressure response, whereas the late time pressure response is not highly affected.
- **3. As** the frictional effects increase, the durations of significant frictional head losses decrease.
- **4.** Early time responses in the presence of wellbore skin differ significantly from early time responses without wellbore **skin**.
- 5. The flowrate profile in the presence of high wellbore skin is practically constant, and decreases rapidly like a shut-in. Hence, the late time build-up with high wellbore skin could be approximated as a simple build-up flowing a constant rate flow period, and analyzed using a conventional Horner technique.

# **Recommendations for Future Studies**

- 1. Consider the effect of momentum effects on the closed chamber well test.
- **2.** Analyze the influence of the produced liquid compressibility on the closed chamber pressure response.
- **3.** Improve the superposition model with variable time steps by including a restarting routine.

## NOMENCLATURE

A <sub>ch</sub> =	Cross sectional	area of the	chamber	$(ft^2)$
				~ .

- $c_t =$  **Total** formation compressibility  $(p_{s_t})^{-1}$
- D = Inside pipe diameter (ft)
- $D_{ch}$  = Chamber diameter (ft)
  - *e* = Absolute roughness (*in*)

e/D = Relative roughness

- f = Moody friction factor
- g = Acceleration of gravity constant (32.2 ft/sec<sup>2</sup>)
- $g_c =$  Conversion factor (32.2 *lbm-ft/lbf-sec*<sup>2</sup>)
- h = Formation thickness (*ft*)
- $h_f =$  Frictional head loss (ft)
- k = Reservoir permeability (*mD*)
- $K_0$  = Modified Bessel function of second kind, zero order
- $K_1$  = Modified Bessel function of second kind, first order
- L = Pipe length (ft)
- $L_c =$  Total chamber length, as illustrated in Figure 2 (*ft*)
- $L_{ci}$  = Initial fluid level, as illustrated in Figure 2 (ft)
- N = Time step index
- $N_p$  = Cumulative liquid production ( $fr^3$ )
- $p_{ch}$  = Chamber pressure (*psia*)
- $p_{ch_i}$  = Initial chamber pressure (*psia*)
- $p_1 =$  Pressure at point 1, as expressed in Equation (1) (*psi*)
- $p_2 =$  Pressure at point 2, as expressed in Equation (1) (*psi*)
- $p_D$  = Dimensionless pressure
- $p_i$  = Static initial reservoir pressure (*psia*)

Po =	Minimum well bore pressure achieved during the test (psia)
$p_{wf} =$	Bottom hole flowing pressure (psia)
$\Delta p =$	Pressure change (psi)
<i>q</i> =	Flowrate (ft <sup>3</sup> /sec)
$q_o =$	Oil flowrate ( <i>bbl/D</i> )
<i>Q</i> =	Cumulative influx (bbl)
<i>QD</i> =	Dimensionless cumulative influx
$\overline{Q}_D =$	Laplace Dimensionless cumulative influx
<i>r</i> =	Radius (ft)
$r_w =$	Wellbore radius (ft)
$r_D =$	Dimensionless radius
Re =	Reynolds number
<i>s</i> =	Laplace variable
<i>S</i> =	Dimensionless skin factor
<i>t</i> =	Time (sec)
At =	Time step (sec)
$t_D =$	Dimensionless time
$\Delta t_D =$	Dimensionless time step
v =	Specific volume (( <i>lbm/ft</i> <sup>3</sup> ) <sup>-1</sup> )
V =	Average fluid velocity (ft/sec)
$\Delta(V^2) =$	Change in velocity terms (ft <sup>2</sup> /sec <sup>2</sup> )
$V_{ch} =$	Chamber gas volume $(ft^3)$
$V_{ch_i} =$	Initial chamber <b>gas</b> volume $(ft^3)$
<i>X</i> =	Dynamic fluid level, as illustrated in Figure 2 (ft)
A x =	Fluid level change <b>per</b> time step ( <i>ft</i> )
<i>z</i> =	Elevation (ft)
$\Delta z =$	Change in elevation ( <i>ft</i> )

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- Z = Real gas compressibility factor of chamber gas
- $Z_i$  = Initial real gas compressibility factor of chamber gas
- **a** = Kinetic energy correction term
- $\beta$  = Influx constant (*bbl/psi*)
- $\boldsymbol{\mu} = \quad Fluid \text{ viscosity } (cp)$
- $\mu_o =$  Oil viscosity (cp)
- $\phi$  = Formation porosity (fraction)
- $\rho_{f=}$  Fluid density (*lbm/ft*<sup>3</sup>)

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# APPENDIX 1 COMPUTER PROGRAM

In this appendix the computer program is presented. Since the program for constant time steps is a subset of the program for variable time steps, only the latter one is presented. Also, the program output for the basecase is presented following the computer code.

CL	OSED CHAMBER WELL TEST RESPONSE SIMULATOR
	(FOR VARIABLE TIME STEPS)
*******	*********
This progra	am generates a pressure as a function of time or a Closed Chamber Test. The method of <b>solution</b>
1s a wellb	ore material balance of the fluids produced.
pressure C	umulative influx solution.
The dimens	fonless cumulative influx value is calculated by
for differ	ent dimensionless time values are in a file
called 'ta Run parame	ble". eters are read from a file called 'basedata". The
output fil	e name is also specified in the data read from
The progra	m works interactively, time steps and number of
time steps	for a given time step are chosen according to the
developmen	it of the pressure response.
Variable D	efinitions:
	Closed Chamber Area (6+\$\$2)
ALC =	Total Chamber Length as shown in figure 2 (ft)
ALI =	Initial Fluid Column Length (ft) Broducod Liquid Convitu (Dogroos API)
B =	Influx Constant ((ft**3)/ps1)
CDTD =	Constant for Evaluation of the Dimensionless
CFH =	Constant for Evaluation of the Frictional Head
CFP =	Loss Constant for Evaluation of the Fluid Produced
CPSI =	Constant for Evaluation of Pressure Drops
CQO =	Constant for Evaluation of the 011 Flowrate
	Fluid Column Length within chamber (ft)
CT =	Total Formation Compressibility (ps(**(-1))
DB ≢	Lower Surge Valve Depth (ft)
DES =	Logical Variable (y/n)
DF = DT =	Time Step (seconds)
DU =	Upper Surgè Valve Depth (ft)
DVV = DZ =	Change in <b>Z</b> during one iteration
	Absolute Roughness (in)
ED ■ F ■	Moody Friction Factor
FH ■	Frictional Head Loss (ft)
FP1 =	Cumulative Liquid Produced (BBT) Cumulative Liquid Produced for the previous
<u> </u>	time step (bbi) Chember Cos Fraulty (relative to all)
uu = H =	Formation Thickness (ft)
	<ul> <li>Elad Variable for Iterative Z factor Calculation</li> </ul>
IFLAG •	
IFLAG •	0 = Convergence Achteved 1 = No Convergence

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0 = 00 greater than zero \* 1 = 00 less or equal to zero Total Number of Time Steps Cumulative Number of Time Steps before fnput N = \* -N1 \* \* a new time step Number of Time Steps for a given time step N2 = -\* NAME = Output File Name . \* Counter of Time Steps for a given time step NCOUNT = • Number of Output Data points . NDP = \* NLOU = × Closest Lower Entry in the Table \* NOUT = Number of Time Steps between output data points \* \* P = Bottom Hole Flowing Pressure Array (psia) \* PC = \* Pseudo Critical Pressure (psia) . PCH = Chamber Gas Pressure (psia) Chamber Gas Pressure for the previous ttme step Initial Chamber Gas Pressure (psig)--->(psia) . × PCHI = \* \* PCHI = \* × Hydrostatic Pressure Drop (psi) Formatton Permeability (milli-darcy) PELEV = ٠ PERM = \* Frictional Pressure Drop (psi) PFRIC = Frictional Pressure Drop for the previous time step \* PFRICI = Formation Porosity (fraction) Initial Reservoir Static Pressure (psig)--->(psia) PHI = \* P I = \* PR = Pseudo Reduced Pressure • Dimensionless Cumulative Influx Value Obtained qddummy = • \* from Interpolation Dimensionless Cumulative Influx Values in the qdtab = \* "table" for Interpolation Dimensionless Cumulative Fluid Influx Array 011 Flowrate (bb1/D) × Q = \* \* Q0 = \* 001 = C Flowrate for the Previous Time Step (bb1/D) \* 0il Flowrate for the Previous Time Step (bb1/D) × QOTEM = \* RF = Reynolds Number \* Wellbore Radius (ft) Specific Liquid Gravity (relative to water) \* RW = . SGF = • \* . SKN = Hurst Skin Factor × tddummy # Dimensionless Time \$ tdtab = Dimensionless Time Values in the Table for Interpolation Time Elapsed Array (seconds) Time fn which the Pressure Drop **is in** Effect \* Τ = Ţ \* TI = Array (seconds) Pseudo Critical Temperature (R) × TC = Frictional Pressure Drop for the previous time step\* Chamber Gas Temperature (F)--->(R) \* TEM = • TEMP = TR = Pseudo Reduced Temperature Total **Time** (seconds) × TT = • Total Time for the previous time step TT1 = Fluid Level Measured from Mid-Perforations (ft) \* UF = \* Ζ= Chamber Real Gas CompressIbility Factor Initial Chamber Real Gas 'CompressIbility Factor •  $\overline{ZI} =$ \* Input variable units are "field units" and converted to NOTE: absolute units during execution. In the above variable listing this change in units in denoted as: (input unit)--->(output unit) \* \* \* • \* \* \* SUBROUTINES: Moody Friction Factors Calculation. Pseudo Critical Temperature and Pressure \* FM × ş, • GPC = • Calculation for Hydrocarbon Gas. • GZ = Real Gas Deviation Factor Calculation

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\* for Hydrocarbon Gas. 00000000000000000 • Interpolator to Calculate Cumulative Influx Dimensionless Values. INTERP = • \* • \* \* \* ₩ \* Beatriz Salas (After Simmons, 1985) 튶 • \* August, 1986 \* \*\*\*\*\*\*\* \*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\* IMPLICIT REAL\*8(A-H,O-Z) DIMENSION T(18882), P(18881), Q(18882), T1(18882) CHARACTER\*18 NAME CHARACTER\*1 DES, DES1, DES2, DES3, DES4 DIMENSION qdtab(550),tdtab(550) OPEN(UNIT=2,FILE='basedata',STATUS='OLD',ACCESS='SEQUENTIAL') REWIND(UNIT=2) OPEN(UNIT=3,FILE='table',STATUS='OLD',ACCESS='SEQUENTIAL') REWIND(UNIT=3) OPEN(UNIT=4,FILE='output',STATUS='NEW',ACCESS='SEQUENTIAL') 00000 \* \*\*\*\*\*\*\* Read Run Parameters from a file called 'basedata" \*\*\*\*\*\*\*\* \*\*\*\*\*\*\*\* READ(2,\*)NAME READ(2,\*)D READ(2,\*)E READ(2,\*)DB READ(2,\*)DB READ(2,\*)DU READ(2,\*)DF READ(2,\*)CL READ(2,\*)DF READ(2,\*)CL READ(2,\*)CCHI READ(2,\*)GG READ(2,\*)TEMP READ(2,\*)TEMP READ(2,\*)PHI READ(2,\*)PHI READ(2,\*)SKN READ(2,\*)W READ(2,\*)H READ(2,\*)CT READ(2,\*)API READ(2,\*)API READ(2,\*)UF 000000 Read dimensionless time and dimensionless cumulative lnflux values from a file called "table" ---------DO 5 I=1,55Ø READ(3,\*)tdtab(I),qdtab(I) 5 CONTINUE 0000000 \*\*\*\*\*\*\*\*\*\*\*\*\*\* \*\*\*\*\*\*\*\*\*\*\*\* Change unlts to useable form, calculate initial conditions and constants ED = E/D

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```
D = D/12.
ALC = DF - DU
ALI = DF - DB + CL
         PCHI = PCHI + 14.696
         PI = PI + 14.696
        \begin{array}{rcl} \text{TEMP} &= & \text{TEMP} &+ & 460 \\ \text{RW} &= & & \text{DW}/24 \end{array}
         SGF = 141.5/(131.5+API)
         ACH = (3.1415926)*(D**2)/4
C
         CALL GPC(GG,TC,PC)
         TR = TEMP/TC
         PR = PCHI/PC
         CALL GZ(TR, PR, ZI)
C
         T(1) = 0.0
         P(1) = PCHI + \emptyset.4333*ALI*SGF
         Q(B) = O.O
         B = 1.119*PHI*CT*(RW**2)*H
         CDTD = 73.25E-9 *PERM/(PHI*UF*CT*(RW**2))
         CQO = 86400
         CRE = 1086.08/(UF*D*(131.5+API))
         CFH = 4.22E-9*8/((3.1415926**2)*(D**5)*32.174)
         CPSI = Ø.4333*SGF
         CFP = 5.615/ACH
         PELEV = CPSI*ALI
CCCCC
             *****
                                          Output the data check
         *****************
         OPEN(1,FILE=NAME)
 WRITE(1,3000)
3000 FORMAT('CLOSED CHAMBER WELL TEST (INCLUDING FRICTIONAL EFFECTS)',
&/.105('*'),//,
&'INPUT DATA AS FOLLOWS:',//)

 WRITE(1,3005)NAME
3005 FORMAT('OUTPUT FILE NAME = ',T49,A10,/)
 WRITE(1,3010)D

3010 FORMAT('CHAMBER DIAMETER = ',T47,F6.3,' (FEET)')

WRITE(1,3015)E

3015 FORMAT('ROUGHNESS = ',T46,F7.5,' (INCHES)')
 WRITE(1,3016)ED
3016 FORMAT('RELATIVE ROUGHNESS = ',T46,F7.5)
 3016 FORMAT('RELATIVE ROUGHNESS = ',146,F7.5)
WRITE(1,3020)ALC
3020 FORMAT('TOTAL CHAMBER LENGTH FROM PERFORATIONS = ',
&T45,F8.2,' (FEET)')
WRITE(1,3030)ALI
3030 FORMAT('INITIAL FLUID COLUMN LENGTH = ',T45,F8.2,' (FEET)')
WRITE(1,3040)PCHI
3040 FORMAT('INITIAL CHAMBER PRESSURE = ',T45,F8.2,' (PSIA)')
WRITE(1,2045)CC
  WRITE(1,3045)GG
3045 FORMAT('CHAMBER GAS GRAVITY = ',T47,F6.4,' (AIR=1.0)',/)
  WRITE(1,3050)PI
3050 FORMAT('INITIAL
                                RESERVOIR PRESSURE = ', T45, F8.2, ' (PSIA)')
  WRITE(1,3060)TEMP
3060 FORMAT('RESERVOIR TEMPERATURE = ',T45,F8.2,' (R)')
  WRITE(1,3070)SGF
3070 FORMAT('PRODUCED FLUID SPECIFIC GRAVITY = 'T47.F6.4)
  WRITE(1,3075)UF
3075 FORMAT('PRODUCED FLUID VISCOSITY = ',T47,F6.3,' (CP)')
  WRITE(1,3080)PHI
3080 FORMAT('RESERVOIR POROSITY = ',T47,F6.4)
         WRITE(1,3090)PERM
```

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```
LT43,E10.4,' (1/PSI)')

WRITE(1,3120)H

3120 FORMAT('FORMATION THICKNESS = ',T45,F8.2,' (FEET)',)

WRITE(1.3125)
  3125 FORMAT(4(/), 'NUMBER OF DATA POINTS = ', T50, '100')
 3125 FORMAT(4(/), 'NUMBER OF DATA POINTS = ', T50, '100')
WRITE(1,3130)
3130 FORMAT(//, 'PRESSURE VS TIME DATA:',/,105('*'),//,
&5X, 'TIME', T17.'Pwf', T25.'FLD PRD'.T37, 'PELEV', T47,
&'Pch', T57, 'Z', T63.'IFLAG'.T72.'Re', T81, 'f', T91, 'qo',
&T101, 'PFRIC',/,T3.'(SECONDS)'.
&T15, '(PSIA)', T26, '(BBLS)', T35, '(PSI)', T45,
&'(PSIA)', T89, '(BBL/D)', T99.'(PSI)'./)
WRITE(1,3200)0,P(1),0,PELEV,PCHI,ZI,0,0,0,0
С
          C
C
C
C
C
C
                                      Calculate other initial conditions
          ************
                                                                                **********************
          \mathbf{Z} = \mathbf{Z}\mathbf{I}
          KFLAG=Ø
          FP1 = 0.00
Q01 = 0.00
          PFRIC1 = 0.00
          TT1 = 0.00
          N1 = O
          TEM = 100.00
QOTEM- 100.00
          PCH1 = PCHI
C
C
C
C
C
C
          *****
                                                   Input run data
          *****
C
 40 WRITE(6,5000)
5000 FORMAT('TIME STEP? (SECONDS)')
 5000 FORMAT('IIME SIEFF (SECONDS)',
READ(5,*)DT
WRITE(6.5010)
5010 FORMAT('NUMBER OF TIME STEPS? (INTEGER)')
         READ(5,*)N2
IF(N2.LT.3)THEN
DES2 = 'y'
             GO TO 38
          ENDIF
 ENDIF

WRITE(6,5015)

5015 FORMAT('DO YOU WANT TO CHECK EACH TIME STEP? (y/n)')

READ(5,*)DES2

IF(DES2.EQ.'y')GO TO 38

WRITE(6,5016)

5016 FORMAT('DO YOU WANT TO WRITE THIS SET OF DATA POINTS ?')

READ(5,*)DES3

IF(DES3.EQ.'n')GO TO 38

UPITE(6.5020)
 WRITE(6,5020)

5020 FORMAT('NUMBER OF DATA POINTS ?')

READ(5,*)NDP

NOUT = N2/NDP
С
Ċ
          **************
```

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```
Calculate the Pressure Response
00000
         ********** Calculate Total Number of Time Steps ****************
 38
        = N1 + N2 + \emptyset.1
C
      NCOUNT = Ø
C
C
         C
      DO 10^{\circ} I=(N1+1),N
        NCOUNT = NCOUNT + 1
        IFLAG=Ø
        T(1+1) = T(1) + DT
000000
      *****
            Interpolation to Calculate Dimensionless Cumulative
                              Influx Values
      *******
                           *******
                                                                *********
        DO 11 L=1,I
T1(L) = T1(L) + DT
tddumy = T1(L)*CDTD
CALL INTERP(tddumy,qddumy,qdtab,tdtab)
Q(L) = qddumy
CONTINUE
 11
C C C C C
         Superposition to Determine Cumulative Influx
                   ********
        FP = B \star (PI - P(1)) \star Q(1)
        IF(I.EQ.1)GO TO 24
C
        DO 20 J=2.I
FP = FP = B*(P(J)-P(J-1))*Q(J)
 20
        CONTINUE
C
C
C
24
         ******************* Calculate the Oil Flowrate *********************************
        IF(KFLAG.EQ.1)GO TO 25
OO = CQO*(FP-FP1)/(T(I+1)-T(I))
FP1 = FP
С
   25
        CONTINUE
C
C
C
      **************** Calculate the Dynamic Fluid Level ************************
        X = ALI + CFP * FP
C
CCCC
                         ************************
                                                         ************
                   Iterative Calculation of PCH and Z
        DO 30 L=1,500
PCH = PCHI*(ALC-ALI)*Z/((ALC-X)*ZI)
          PR = PCH/PC
CALL GZ(TR,PR,ZN)
DZ=SQRT((ZN-Z)**2)
          Z=ZN
IF(DZ,LT.Ø.ØØØ1)GO TO 35
   30
        CONTINUE
C
        IFLAG = 1
```

```
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```

35 CONTINUE С \*\*\*\*\*\* The oil flowrate and frictional pressure drop values are checked. If 00 is less or equal to zero or FH less than 0.10 ft, the oil flowrate, Reynolds number, Moody friction factor and frictional head loss are set equal to zero. IF (QOTEM.LE.Ø.ØØ.OR.TEM.LT.Ø.1Ø) THEN QO = Ø.ØØ FH = Ø.ØØ RE = Ø.ØØ f = Ø.ØØ KFLAG = 1 GO TO 36 ENDIF 00000000 The relative roughness value is checked. If ED is equal to zero there are no frictional effects in the tubing, and the frictional head loss is set equal to zero. IF(ED.EQ.Ø)THEN FH=Ø.ØØ GO TO 36 ENDIF C C C \*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\* Calculate Reynolds Number \* RE = CRE \* QOC C C CALL FM(RE.ED.F) С С С \*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\* Calculate Frictional Head Loss \* FH = CFH + F + X + (QO + + 2)CONTINUE 36 С C C C C C C ........ Calculate pressure drops and bottom hole pressure (as expressed in Equation (29)). \*\*\*\*\*\*\*\*\*\*\*\*\* PELEV = CPSI\*X PFRIC = CPSI\*FH P(I+1) = PCH + PELEV + PFRIC CCCCC \* \*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\* Print data check IF(DES2.EQ.'n')GO TO 37 WRITE(6.5025) T(I+1),P(I),P(I+1),QO1,QO,PFRIC1,PFRIC, WRITE(6,5025) ((1+1),P(1),P(1+1),d01,d0,PPRICI, PCH1,PCH FORMAT('TIME = ',F10.5,/, 'P(I) = ',F10.2,2X,'P(I+1) = ',F10.2,/, 'QO(I) = ',F10.2,2X,'QO(I+1) = ',F10.2,/, 'PFRIC(I) = ',F10.2,2X,'PFRIC(I+1) = ',F10.2,/, 'PCH(I) = ',F10.2,2X,'PCH(I+1) = ',F10.2) а 5025 а а а а

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```
00000
        *********
        Ask to repeat last tlme step
      **
        WRITE(6,5026)
FORMAT('DO YOU WANT TO REPEAT THE LAST TIME STEP ? (y/n)')
 5026
        READ(5,*) DES1
C
C
C
      ******** Correct counters if the last time step is repeated *******
        IF (DES1.EQ.'y') THEN
          N = N1 + NCOUNT = N1 = N
                            1
          DO 12 L=1.I
T1(L) = T1(L) = DT
 12
          CONTINUE
          GO TO 40
        ELSE
          DES3 = 'a'
        ENDIF
C
C
      ************* Actualize cumulative data varlablcs *******************
С
 37
        Q01 = Q0
        QOTEM = QO
FP1 = FP
        PFRIC1 = PFRIC
        PCH1 = PCH
        TEH = PFRIC
С
      WRITE(4,3288)T(I+1),P(I+1),FP,PELEV,PCH,Z,IFLAG,RE,F,QO,PFRIC
00000
                                                       ***************
         Selective Data Output
                                                    ****************
      IF(DES3.EQ.'n')THEN
        GO TO 10
      ELSE
        IF (DES3.EQ.'y')THEN
          GO TO 39
        ELSE
          IF (DES3.EQ. 'a') THEN
            WRITE(6.5027)
FORMAT('DO YOU WANT TO WRITE THIS DATA POINT ?')
READ(5.*)DES4
IF(DES4.EQ.'n')THEN
 5027
              GO TO 10
            ELSE
              NOUT=1
            GO TO 39
ENDIF
          ELSE
          GO TO 10
ENDIF
        ENDIF
      END I F
С
 39
        RI = I
        AA = RI/NOUT
        NA = AA
BB = NA
        IF (AA.NE.BB)GO TO 10
```

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```
С
     WRITE(1,3288)T(I+1),P(I+1),FP,PELEV,PCH,Z,IFLAG,RE,F,QO,PFRIC
 3200 FORMAT(F10.5,F10.2,F10.4,2F10.2,F10.4,15,F12.3,2X,
    &F5.3,F1Ø.2,F1Ø.2)
   10 CONTINUE
0000
      **********
                             Ask to Contfnue
          ******
Ĉ
     WRITE(6,5025)T(N+1),P(N),P(N+1),Q01,Q0,PFRIC1,PFRIC,PCH1,PCH
 WRITE(6,5030)
5030 FORMAT('DO YOU WANT TO CONTINUE ? (y/n)')
     READ(5,*)DES
C
      IF (DES.EQ.'y') THEN
       N1 = N
       GO TO 40
      ELSE
       WRITE(6,5Ø35)
5035 FORMAT('ARE YOU SURE?')
READ(5,*)DES
IF(DES.EQ.'n')THEN
         N1 = N
GO TO 40
        ENDIF
      ENDIF
CCCC
      STOP
      END
C
C
C
C
C
C
C
      *****************
      SUBROUTINE FM(RE,ED,F)
C
C
C
      ******************
      THIS SUBROUTINE CALCULATES THE MOODY FRICTION FACTOR GIVEN THE REYNOLDS NUMBER AND THE
0000000
      RELATIVE ROUGHNESS. THE EQUATIONS USED ARE
THOSE PRESENTED IN SECTION 3.2. (EQUATIONS
      (12), (13), (14) AND (15)).
                             **********
      *******
С
      IMPLICIT REAL*8(A-H,O-Z)
C
      SL = (200/ED)**1.16
C
      IF (RE.LE.2000) THEN
      F = 64/RE
      ELSE
        IF (RE.LE.4000) THEN
F = 0.5/(RE**0.3)
        ELSE
          IF (RE.LE.SL) THEN
          \overline{F1} = ED
          DO 40 J=1,188
          Y = ED+9.34/(RE*DSQRT(F1))
```

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```
F = (1.14-2*DLOG18(Y))**-2
DIF = DABS(F-F1)
        IF (DIF.LT.1E-6) GO TO 45
F1 = F
 40
         CONTINUE
         ELSE
          F = (1.14-2*DLOG1Ø(ED))**-2
        ENDIF
       ENDIF
     ENDIF
C
  45
    RETURN
     END
C
C
C
C
     ***************
                        Ċ
     SUBROUTINE GPC(GG, TC, PC)
C
C
     C
C
C
     THIS ROUTINE CALCULATES THE PSEUDOCRITICAL
     TEMPERATURE AND PRESSURE FOR CONDENSATE WELL
00000
     FLUIDS, GIVEN THE GAS GRAVITY. THE EQUATIONS
     USED ARE THOSE GIVEN BY STANDING.
     IMPLICIT REAL*8(A-H,O-Z)
С
     PC = 706. - 51.7*GG - 11.1*(GG**2)
TC = 187 + 33Ø*GG - 71.5*(GG**2)
C
     RETURN
     END
C
C
C
     **********************
Ĉ
     SUBROUTINE GZ(TR, PR, Z)
C
C
C
     ***********************
C
C
     THIS ROUTINE CALCULATES THE GAS DEVIATION
     FACTOR FOR A NATURAL GAS GIVEN THE REDUCED
     PSEUDOCRITICALS TEMPERATURE AND PRESSURE.
C
C
C
     THE EQUATIONS USED ARE CURVE FIT RELATIONS FROM THE STANDING-KAT2 CHART GIVEN BY
C
     BRILL AND BEGGS.
Ĉ
C
C
     *****
     IMPLICIT REAL*8(A-H,O-Z)
C
     A = 1.39*SQRT(TR-Ø.92) - Ø.35*TR - 0.101
C
    B = {Ø.62 - Ø.23*TR}*PR
&+ {Ø.Ø66/(TR-Ø.86} - Ø.Ø37)*(PR**2}
&+ {Ø.32/(1Ø**(9*(TR-1))))*(PR**6)
C
     C = (0.132 - Ø.32*DLOG1Ø(TR))
C
```

```
D =18.**(8.3186 - 8.49*TR +8.1824*(TR**2))
С
      Z = A + (1-A)/(2.718281828**B) + C*(PR**D)
С
      RETURN
      END
С
Č
C
č
C
      ********
      SUBROUTINE INTERP(tddumy,qddumy,qdtab,tdtab)
000000000
      ****
      THIS ROUTINE INTERPOLATES BETWEEN TWO SUCCESSIVE
VALUES IN A TABLE THAT CONTAINS THE DIMENSIONLESS
CUMULATIVE INFLUX VALUES FOR DIFFERENT DIMENSIONLESS
      TIME VALUES.
      FOR A GIVEN DIMENSIONLESS TIME THE INTERPOLATOR
SUBROUTINE COMPUTES THE CLOSEST LOWER ENTRY IN A
TABLE, AND INTERPOLATES BETWEEN THIS VALUES AND
C C C C C
      THE FOLLOWING ENTRY IN THE TABLE.
                                *********
Č
      IMPLICIT REAL*8 (A-H,O-Z)
      dimension tdtab(55Ø),qdtab(55Ø)
С
      if (tddumy.lt.\emptyset.1) go to 1\emptyset
if (tddumy.lt.1.\emptyset) go to 2\emptyset
if (tddumy.lt.1\emptyset.\emptyset) go to 3\emptyset
if (tddumy.lt.1\emptyset.\emptyset) go to 4\emptyset
if (tddumy.lt.1\emptyset\emptyset.\emptyset) go to 5\emptyset
С
Č
      **************
      nlow=46Ø+(tddumy-1ØØØ.Ø)/1ØØ.Ø
      qddumy=qdtab(nlow)+(qdtab(nlow+1)-qdtab(nlow))*
     & (tddumy-tdtab(nlow))/188.8
      go to 999
C
C
      С
5Ø
      nlow=37Ø+(tddumy-1ØØ.Ø)/1Ø.Ø
      qddumy=qdtab(nlow)+(qdtab(nlow+1)-qdtab(nlow))*
     & (tddumy-tdtab(nlow))/10.0
      go to 999
С
      Č
                                            *********
Ċ
40
      n low=28\emptyset+(tddumy-1\emptyset.\emptyset)
      qddumy=qdtab(nlow)+(qdtab(nlow+1)-qdtab(nlow))*
     & (tddumy-tdtab(nlow))/1.Ø
      go to 999
С
С
      Ċ
30
      nlow=19\%+(tddumy-1.\%)*1\%
      qddumy=qdtab(nlow)+(qdtab(nlow+1)-qdtab(nlow))*
     & (tddumy-tdtab(nlow))/Ø.1
      go to 999
С
С
```

```
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```

ı

#### APPENDIX 2

## NUMERICAL VALUES FOR THE BASECASE WITH AND WITHOUT FRICTION

CLOSED CHAMBER WELL TEST (INCLUDING FRICTIONAL EFFECTS)

INPUT DATA AS FOLLOWS:

OUTPUT FILE NAME # 0.203 (FEET) 0. (INCHES) CHAMBER DIAMETER = ROUGHNESS = Ο. RELATIVE ROUGHNESS = 0. TOTAL CHAMBER LENGTH FROM PERFORATIONS = 1000.00 (FEET) 0. (FEET) 44.70 (PSIA) 0.650 (AIR=1.Ø) INITIAL FLUID COLUMN LENGTH = INITIAL CHAMBER PRESSURE = CHAMBER GAS GRAVITY = INITIAL RESERVOIR PRESSURE = 5014.70 (PSIA) 635.00 (R) RESERVOIR TEMPERATURE = PRODUCED FLUID SPECIFIC GRAVITY = PRODUCED FLUID VISCOSITY = 0.9042 1.250 (CP) 0.2700 RESERVOIR POROSITY = RESERVOIR PERMEABILITY = 100.00 (MD) SKIN = 0. 0.8333 (FEET) Ø.1000e-04 (1/PSI) 25.00 (FEET) WELL DIAMETER = FORMATION TOTAL COMPRESSIBILITY = FORMATION THICKNESS = NUMBER OF DATA POINTS = 87

#### PRESSURE VS TIME DATA:

Pch Z Re (PSIA) TIME P**wf** FLD PRD PELEV (SECONDS) **(PSIA) (BBLS)** (PSI) f PFRI qo 

 (PSIA)

 44.70
 0.9964
 0.
 0.

 49.50
 0.9960
 0.
 0.

 53.34
 0.9957
 0.
 0.

 60.01
 0.0254
 0.
 0.

 (BBL/D) (PSI *0.* 0.5640 38.17 0.9411 63.70 1.2787 0. 1.**0**0000 44.70 Ο. ο. 87.68 0. Ε. 0. 0. 2.00000 Ø. 117.04 3.00000 143.87 0. **0**. *0*. 0. 4.00000 169.48 1.5938 107.88 61.60 0.9950 Ο. *e* . Ò. 5.00000 1.8933 128.15 66.31 0.9946 Ø. 194.46 Ο. 147.62 71.56 0.9942 ø. 6.00000 219.18 2.1808 Ο. Ο. ø. 243.93 0.9937 0. Ο. 2.4587 166.43 77.50 Ο. Ø. 7.00000 84.29 Ο. Ø. 8.00000 268.98 2.7285 184.69 0.9932 Ο. 0. Ο. 0. 9.00000 294.59 2.9909 202.45 92.14 0.9925 Ο. Ο. Ο. 10.00000 321.16 3.2472 219.80 101.36 0.9918 0. ο. ο. 0. 236.78 3.4980 Ο. 11.00000 349.14 112.36 0.9909 Ο. 0. 0.9898 12.00000 3.7434 253.39 125.70 Ο. Ο. 0. 379.09 0. Ο. 13.00000 411.83 3.9832 269.62 142.21 0.9885 Ο. Ο. ο. 0. Ø. 14.00000 448.71 4.2180 285.51 163.20 0.9867 Ο. ø. Ο. 4.4476 190.72 0.9845 Ο. Ο. 15.00000 491.78 301.06 0. 0.9814 228.28 544.50 4.6717 316.22 Ο. 0. 16.00000 Ο. 282.14 Ο. 4.8889 Ο. Ο. 17.88888 613.06 330.92 0.9769 Ο. 0. 0. Ø. £. 18.00000 710.08 5.0979 345.07 365.01 0.9699 ο. Ο. 0. Ο. 19.00000 863.46 5.2952 350.43 505.03 0.9582 Ο. 0. 0. 1142.34 1702.33 0. 20.00000 5.4729 370.46 771.88 0.9361 Ο. 5.6120 379.87 1322.46 0. 0.8956 Ο. 21.00000 0. 22.00000 384.78 Ο. 0. 2545.07 5.6846 2160.28 0.8588 0. 0. 0. 0. 0. Ø. 23.00000 3116.22 5.7065 386.27 2729.96 0.8550 24.00000 3419.51 5.7142 386.79 3032.72 0.8596 0. 5.7181 5.7205 387.05 387.22 Ó. 3609.31 0. 25.00000 3222.26 0.8648 0.8693 0.8712 26.00000 3742.4 1 3355.19 Ο. ο. 26.42000 3792.88 5.7214 387.27 Ø. 0. 3405.61

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basecase

\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*

27 42000	2800 00	5 7 2 2 0	207 26	2502 62	0 0754	•	•	•	_
27.42000	3690.00	5.7229	387.30	3502.62	0.8751	Ο.	Ο.	Ο.	5.
28.42000	3971.80	5.7242	387.46	3564.34	0.8787	0.	0.	0	0.
29.4 2000	4041.62	5 7 2 5 1	387 53	3654 09	0 8610	0	<u> </u>	<u>,</u>	
20 4 2000	4400 70	5.7251	307.55	3034.03	0.0019	υ.	0.	υ.	ĸ.
30.42000	4100.79	5.7259	387.56	3713.20	0.8848	Ο.	Ο.	Ο.	ο.
31.55000	4157.94	5.7267	387.63	3770.31	0 8877	0	0	ñ	p
32 55000	1202 17	5 7 2 7 2	207 67	2615 50	0.0011	<u>.</u>	<i>.</i>	0.	ι.
32.55000	4203.17	5.7272	307.07	3615.50	0.8901	υ.	Ο.	Ο.	P.
33.55000	4243.92	5.7277	387.70	3856.22	0.8923	0	0.	0	P
34 55000	4276 60	5 7281	397 73	2000 07	0 0010	<u>.</u>	<u>.</u>	<u> </u>	<i>.</i>
34.33000	4270.00	5.7201	307.73	3000.07	0.8940	υ.	υ.	υ.	Ο.
35.55000	4309.45	5.7285	387.76	3921.70	0.8959	0.	0.	0.	0.
36 55000	4339 61	5 7 2 8 8	397 79	2051 82	0 9076	<u> </u>	<u> </u>	<u> </u>	
00.00000	4000.01	5.7200	507.70	3331.03	0.0970	0.	υ.	υ.	ν.
37.55000	4367.56	5.7291	387.80	3979.76	0.8992	ø.	0.	0.	е.
38.55000	4389.56	5.7294	387 82	4001 76	0 0005	0	ñ	<u> </u>	r ·
20 55000	4440.40	5 7000	007.02	4001.70	0.3003	<i>v</i> .	0.	0.	<i>E</i> .
39.55000	4412.42	5.7296	387.83	4024.58	0.9018	Ο.	Ο.	Ο.	Ø.
40.55000	4434.22	5.7299	387.85	4046.38	0.9031	0	0	0	6
11 55000	1151 77	5 7201	207 06	1066 01	0.0044	<u>.</u>	<u>.</u>	<u>.</u> .	~.
41.55000	4434.77	5.7301	387.80	4000.71	0.9044	υ.	Ο.	υ.	е.
42.55000	4471.21	5.7302	387.87	4083.33	0.9054	0	0	0	Ρ.
13 55000	1188 35	5 7304	397 90	4100 46	0 0064	<u> </u>	ă.	<u> </u>	
43.33000	4400.33	5.7304	307.09	4100.40	0.9064	υ.	ю.	υ.	е.
44.55000	4504.92	5.7306	387.90	4117.02	0.9074	0.	0.	0.	5.
45 55000	4520 80	5 7 3 0 7	387 91	4132 90	0 0 0 8 /	õ	<u> </u>	<u> </u>	ã
46.55000	4520.00	5.7007	007.01	4102.00	0.9004	υ.	0.	0.	<i>k</i> <sup>,</sup>
46.55000	4533.27	5.7309	367.92	4145.36	0.9092	Ο.	Ο.	Ο.	е.
47.55000	4546.71	5.7310	387.92	4158.78	0 9100	0	0	0	ø
18 55000	4550 91	5 7 2 4 4	207 02	4171 00	0.0100	<u>.</u>	ă.	<u> </u>	~·
40.55000	4559.01	5.7311	387.93	41/1.00	0.9109	υ.	ю.	υ.	κ.
49.55000	4572.48	5.7312	387.94	4184.54	0.9117	0	0.	0	0
50 55000	1582 32	5 7313	387 05	4104 27	0 0122	<u><u> </u></u>	<u> </u>	<u>.</u>	~`
50.55000	4302.32	5.7515	307.95	4194.37	0.9123	υ.	υ.	υ.	ĸ.
51.55000	4593.06	5.7314	367.95	4205.11	0.9130	Ο.	Ο.	Ο.	Ø.
52.55000	4603.72	5 7315	387 96	4215 76	0 9137	ñ	ñ	Ô	Ø
	404440	5 7040	207.07	4000.00	0.0107	<i>v</i> .	<i>.</i>	<i>v</i> .	
53.55000	4614.18	5.7316	387.97	4226.22	0.9143	Ο.	Ο.	ø.	е.
54.55000	4622.03	5.7317	387.97	4234.06	0.9148	0	0	0	0
55 55000	1630 80	5 7 2 4 9	207 00	4242.04	0 0 1 5 4	<u>.</u>	<u>č</u> .	<u> </u>	<i>.</i>
55.55000	4030.09	5.7310	307.90	4242.91	0.9154	υ.	υ.	υ.	е.
56.55000	4639.79	5.7319	387.98	4251.80	0.9160	0.	0.	0.	0.
57 55000	4648 54	5 7310	387 99	1260 55	0 0166	ñ	<u> </u>	ò	, o
57.55000	4040.04	5.7515	507.55	4200.33	0.9100	υ.	0.	υ.	0.
58.55000	4654.97	5.7320	387.99	4266.98	0.9170	Ο.	ο.	Ο.	£.
59.55000	4662.38	5 7321	388 00	4274 38	0 9175	0	0	Ø	0
61 15000	4674.04	5 7000	200.00	4000.00	0.0170	<u>.</u>	<u>.</u>	~.	
01.15000	4074.04	5.1322	388.00	4286.03	0.9182	υ.	υ.	Ο.	Ο.
62.05000	4679.36	5.7322	388.01	4291.36	0.9186	0.	0.	0.	5.
63 20000	4687 73	5 7323	388 01	1200 72	0 0101	õ	Ň.	ñ	0
00.20000	4007.75	5.7525	300.01	4233.12	0.3131	0.	υ.	0.	0.
64.60000	4694.52	5.7324	388.02	4306.51	0.9196	Ο.	ø.	Ο.	<i>v</i> .
65.10000	4698.56	5.7324	388.02	4310.56	0.9196	0	0.	0	ρ.
65 00000	4702.27	5 7 2 2 4	200 02	4245 25	0 0 0 0 0	<u> </u>	<u> </u>	<u>.</u>	~.
05.90000	4703.27	5.7324	300.02	4315.25	0.9202	υ.	υ.	υ.	υ.
66.10000	4705.07	5.7324	388.02	4317.05	0.9203	0.	ø.	Ο.	5.
67 10000	4710 31	5 7325	386 03	1322 20	0 0 2 0 6	ñ	0	Ô	0
07.10000	4710.51	5.7525	300.03	4322.23	0.9200	0.	υ.	υ.	v.
68.20000	4716.96	5.7325	388.03	4328.95	Ø.9211	0.	Ο.	Ο.	5.
69.00000	4720.05	5.7326	388.03	4332.02	0 9213	0	0	0	5.
70 00000	4705.00	5 7000	200.04	4002.02	0.0210	<u>.</u>	~.	<u>.</u>	2.
10.00000	4725.22	5.7320	388.04	4337.18	0.9216	υ.	ю.	υ.	5.
71.10000	4731.21	5.7327	388.04	4343.17	0.9220	ø.	0.	0.	0.
72 00000	1731 10	5 73 27	388 04	1316 15	0 0 2 2 2	0	õ	õ	à
12.000000	4734.13	5.7527	300.04	4340.15	0.9222	0.	υ.	0.	æ.
74.60000	4747.76	5.7328	388.05	4359.71	0.9231	ø.	Ο.	Ο.	ο.
75 60000	4748 54	5 7328	388 05	4360 49	0 9232	0	0	0	P.
70.00000	4750.04	5.7520	000.00	4000.40	0.5252	<i>v.</i>	<u>.</u>	<u>.</u>	
76.000000	4/52.59	5.7329	388.05	4364.54	0.9235	Ο.	υ.	υ.	е.
77.20000	4755.72	5.7329	388.05	4367.67	0.9237	0.	0.	0.	0.
79 10000	4759 40	5 7220	200.05	4270.20	0 0 0 0 0	<u> </u>	<u> </u>	<u> </u>	0
70.10000	4/ 00.42	5.7329	300.03	43/0.30	0.9239	υ.	υ.	<i>v</i> .	υ.
79.10000	4762.02	5.7329	388.06	4373.96	0.924 1	Ο.	Ο.	ø.	e.
83.35000	4767 55	5 7330	388 06	4379 49	0 9245	0	0	0	0
	4775 00	5.7550	200.00	4067 74	0.0270	<i>.</i> .	<i>.</i> .	ĕ.	ÿ.
92.20000	4//5.80	5.7331	300.00	4301.14	0.9250	υ.	υ.	υ.	υ.
85. <i>00000</i>	4782.57	5.7331	388.07	4394.50	0.9255	0.	ø.	Ο.	Ο.
86 00000	4790 11	5 7332	388 07	1102 04	0 0260	Ő.	0	0	Â
	4700.11	5.1332	500.07	4402.04	0.9200	<i>v</i> .	υ.	<i>.</i> .	<i>v</i> .
87.10000	4798.98	5.7333	388.08	4410.91	0.9266	Ο.	Ο.	Ο.	Ο.
95.33000	4804.13	5.7333	388.08	4416.05	0.9270	0	ø.	0.	5.
06 00000	4040 40	5 7 2 2 4	200 00	4420.04	0 0 0 7 0	č.	~.	ă.	~
30.00000	4010.13	5.1334	300.09	4430.04	0.92/9	υ.	υ.	v.	υ.
101.95000	4820.94	5.7334	388.09	4432.85	0.9281	Ο.	Ο.	Ο.	Е.

### CLOSED CHAMBER WELL TEST [INCLUDING FRICTIONAL EFFECTS)

INPUT DATA AS FOLLOWS:

OUTPUT FILE NAME .	skinø				
CHAMBER DIAMETER * ROUGHNESS * RELATIVE ROUGHNESS = TOTAL CHAMBER LENGTH FROM PERFORATIONS * INITIAL FLUID COLUMN LENGTH * INITIAL CHAMBER PRESSURE * CHAMBER GAS GRAVITY *	0.203 0.00060 0.00025 1000.00 0. 44.70 0.6500	(FEET) (INCHES) (FEET) (FEET) (PSIA) (AIR=1.0)			
INITIAL RESERVOIR PRESSURE * RESERVOIR TEMPERATURE • PRODUCED FLUID SPECIFIC GRAVITY • PRODUCED FLUID VISCOSITY • RESERVOIR POROSITY * RESERVOIR PERMEABILITY * SKIN • WELL DIAMETER • FORMATION TOTAL COMPRESSIBILITY * FORMATION THICKNESS *	5014.70 635.00 0.9042 1.250 0.2700 100.00 0. 0.8333 Ø.1ØØØe-Ø4 25.00	<pre>(PSIA) (R) (CP) (MO) (FEET) (1/PSI) (FEET)</pre>			
NUMBER OF DATA POINTS -	84				

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PRESSURE VS TIME DATA:

TIME (SECONDS)	Pwf (PS[A)	FLD PRD (88LS)	PELEV (PSI)	Pch (PSIA)	Z.	Re	f	q¢ (BBL/D)	PFRIC (PSI)
TIME (SECONDS) 0. 1.80888 1.80000 2.80000 3.80000 5.88888 6.80000 7.80000 9.20000 10.20000 11.20000 11.20000 12.20000 13.20000 14.20000 14.20000 15.20000 16.20000 17.15000 18.20000 19.40000 21.48888 22.40000	Pwi (PS[A) 44.70 283.72 341.04 400.46 451.10 506.16 540.07 579.85 617.59 669.41 706.47 743.15 777.59 809.70 843.99 083.02 920.77 962.99 1018.66 1096.53 1184.95 1323.31 1581.86	FLD PRD (88LS) 0. 0.5442 0.8353 1.1598 1.4588 1.7952 2.0092 2.2675 2.5172 2.8542 3.0877 3.3157 3.5382 3.7557 3.9688 4.1776 4.3817 4.5716 4.7768 5.1858 5.3584 5.5159	PELEV (PSI) 0. 36.83 56.54 78.50 98.74 121.51 136.00 153.49 170.38 193.20 209.01 224.22 268.64 282.78 296.59 309.45 323.34 338.72 351.02 362.71 373.37	Pch (PSIA) 44.70 49.32 52.20 55.85 59.69 64.69 68.33 73.32 78.88 87.87 95.41 104.14 114.34 126.45 141.10 159.16 181.91 209.82 251.52 322.45 416.48 575.81 885.37	Z 0.9964 0.9958 0.9955 0.9955 0.9948 0.9945 0.9945 0.9945 0.9941 0.9936 0.9929 0.9923 0.9923 0.9923 0.9916 0.9907 0.9885 0.9871 0.98871 0.98871 0.98871 0.98871 0.98871 0.98871 0.9852 0.9735 0.9656 0.9522 0.9270	Re 927830.969 809004.698 732644.987 683599.336 643016.519 621905.952 599542.589 580562.927 559060.162 546435.559 535040.940 523206.796 510495.392 499180.416 489994.049 478635.336 468959.733 458386.672 442938.893 425320.050 400657.387 360082.467	I 8. 0.015 0.015 0.015 0.016 0.	<pre>     (BBL/D)</pre>	PFRIC (PSI) 0. 197.57 232.30 266.18 292.67 319.05 335.74 353.04 353.04 353.04 353.04 402.04 414.57 423.75 429.83 434.26 441.09 442.27 443.72 443.80 435.36 417.45 384.80 323.13
23.40000	2152.23	5.6425	381.94	1574.28	0.8809	273387.370	0.017	10016.77	196.61
24.90000 25.90000	3203.59 3437.35	5.7Ø16 5.7137	386.42 386.82	2812.88 3050.42	0.8543 0.8591	33920.492 4243.251	0.020 0.026	1242.83 155.47	<b>4.29</b> Ø.11
26.00000 27.00000	3456.70 3656.32	5.7150 5.7190	386.84 387.11	3069.85 3269.21	0.8605 0.8663	Ø. O.	0. 0.	0. 0.	0. 0.

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27.82000	3771.27	5.7210	387.25	3384.02	0.8703	0.	0.	0.	0.
28.82000	3880.72	5.7228	387.37	3493.35	0.8747	0.	õ	0	0
29.82000	3966.00	5,7241	387.46	3578.54	0.8784	Õ.	ñ.	0	<i>.</i>
30,82000	4038.44	5.7251	387.53	3650 91	0 8818	ŏ.	<u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u><u></u></u>	0.	<i>.</i>
31.82000	4100.74	5 7259	387 58	3713 16	0.0010	0.	<i>.</i>	0.	<i>v</i> .
32 82000	4153 90	5 7266	297 62	2766 27	0.0010	0.	0.	0.	0.
22 02000	4100 22	5.7200	207.03	3/00.2/	0.00/5	0.	0.	0.	0.
33.02000	4190.22	5./2/2	38/.0/	3810.50	0.8898	0.	ю.	0.	ο.
34.82000	4239.21	5.7277	387.70	3851.51	0.8920	0.	Ο.	0.	Ο.
35.82000	4276.54	5.7281	387.73	3888.81	0.8940	0.	Ο.	0.	Ο.
36.82000	4309.98	5.7285	387.76	3922.22	0.8959	Ο.	Ο.	0.	ο.
37.82000	4338.32	5.7288	387.78	3950.55	0.8975	Ο.	Ο.	0.	ο.
38.82000	4365.66	5.7291	387.8Ø	3977.87	8.8991	Ο.	Ο.	Ο.	ο.
39.82000	4391.41	5.7294	387.82	4003.59	0.9006	Ο.	Ο.	0.	5.
40.82000	4414.79	5.7296	387.83	4026.96	Ø.9Ø2Ø	Ö.	Ô.	0.	0.
41.82000	4434.68	5.7299	387.85	4046.83	0.9032	0	ñ	0	0
42.82000	4454.57	5.7301	387.86	4066.71	Ø. 9Ø44	0	õ.	0	<i>.</i>
43.82000	4473.62	5 7303	387 87	4085 74	0 0055	0.	<i>o</i> .	0.	<i>a</i> .
44 82000	4490 99	5 7304	207 00	4102 10	0.9055	0.	0.	0.	<i>v</i> .
45 55000	4502 55	5 7 2 0 4	207.09	4115 66	0.9000	0.	0.	0.	<i>b</i> .
46 55000	4505.55	5,7300	307.90	4110.00	0.9073	0.	0.	0.	0.
40.55000	451/.14	5./30/	387.90	4129.24	0.9082	0.	0.	0.	5.
47.55000	4531.52	5.7308	387.91	4143.60	0.9091	ø.	0.	0.	Ο.
48.55000	4545.58	5.7310	387.92	4157.66	0.9100	Ο.	Ο.	0.	Ο.
49.55000	4559.23	5.7311	387.93	4171.30	0.9108	Ο.	Ο.	0.	ο.
50.55000	4569.82	5.7312	387.94	4181.88	0.9115	Ο.	Ο.	0.	Ο.
51 <b>.55000</b>	4581.19	5.7313	387.95	4193.24	0.9122	0.	0.	0.	0.
52.55000	4592.51	5.7314	387.95	4204.56	0.9129	ø.	Ô.	0.	5.
53.55000	4603.66	5.7315	387.96	4215.69	0.9136	0.	0	0.	0.
54.55000	4612.03	5.7316	387.97	4224.07	0.9142	0	Ő.	0	<i>.</i>
56.00000	4626.63	5.7317	387.98	4238.66	0 9151	а. а	õ.	ŏ.	<i>.</i>
57.00000	4633.95	5.7318	387.98	4245.97	0.9156	<i>.</i> 0	<i>o</i> .	0.	<i>.</i>
57.98000	4642 28	5 7319	387 99	4254 29	0 9161	0.	<i>o</i> .	0.	0.
58 96000	4650 62	5.7319	207.99	1251.23	0.9101	0.	0.	0.	0.
50.90000	4650.02	5.7320	20/.99	4202.03	0.9107	0.	0.	0.	0.
	4030.90	5./320	388.00	42/0.91	0.91/2	0.	0.	0.	0.
61.15000	4666.17	5.7321	388.00	4278.17	0.9177	0.	0.	0.	Ο.
62.00000	4672.16	5.7322	388.00	4284.16	0.9181	0.	Ο.	0.	Ο.
64.00000	4687.48	5.7323	388.01	4299.47	0.9191	Ο.	Ο.	0.	Ο.
65.10000	4691.47	5.7323	388.01	4303.45	0.9194	Ο.	ø.	0.	Ο.
66.20000	4699.3Ø	5.7324	388.02	4311.28	0.9199	Ο.	Ο.	0.	ο.
67.40000	4706.16	5.7325	388.Ø2	4318.14	Ø.92Ø3	Ο.	Ο.	0.	Ο.
68.00000	4708.82	5.7325	388.03	4320.79	Ø.92Ø5	Ο.	0.	0.	Ο.
69.20000	4714.80	5.7325	388.03	4326.77	0.9209	Ö.	Ô.	0.	0.
70.20000	4721.35	5.7326	388.03	4333.31	0.9214	0	Ô.	0.	0.
71.30000	4725.94	5.7326	388.04	4337.91	0.9217	0	ō.	0.	0
71.90000	4730.16	5.7327	388.04	4342.12	8,9220	0	ñ.	0	<i>.</i>
73,10000	4735 10	5 7327	388 04	4347 06	0 9223	0.	<i>o</i> .	0.	<i>.</i>
74 20000	4730 87	5 7328	399 04	4251 92	0 0226	0.	<i>o</i> .	0.	<i>.</i>
75 10000	4742 24	5 7320	300.04	4255 10	0.9220	0.	0.	0.	0.
75.10000	1/1J.41 4746 EQ	5.7520	200.05	4353.19	0.9220	0.	0.	0.	0.
	4740.50	5./328	388.05	4358.40	0.9231	0.	0.	0.	0.
77.50000	4/54.05	5.7329	388.05	4366.60	0.9236	0.	0.	0.	0.
78.30000	4757.60	5.7329	388.05	4369.55	0.9238	0.	0.	0.	Ο.
79.00000	4758.50	5.7329	388.05	4370.45	0.9239	Ο.	Ο.	0.	Ο.
83.45000	4765.97	5.7330	388.06	4377.91	0.9244	Ο.	Ο.	0.	Ο.
84.05000	4773.45	5.7330	388.06	4385.39	0.9249	Ο.	Ο.	0.	Ο.
85.00000	4779.39	5.7331	388.07	4391.32	0.9253	Ο.	Ο.	0.	ο.
86.15000	4789.62	5.7332	388.07	4401.55	0.9260	Ο.	Ο.	0.	Ο.
87.20000	4796.67	5.7332	388. <b>Ø8</b>	4408.59	0.9265	0.	ø.	0.	ο.
95.15000	4802.94	5.7333	388.08	4414.86	0.9269	Ô.	0.	0.	0.
96.00000	4816.57	5.7334	388.09	4428.48	0.9278	Ô.	0	0.	ø.
102.35000	4818.64	5.7334	388.09	4430.56	0.9280	Ö.	<i>0</i> .	Õ.	0.
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